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# New Concept of Telemetry X-Band Circularly Polarized Antenna Payload for CubeSat

Jamil Fouany, Marc Thevenot, Eric Arnaud, François Torres, Cyrille Menudier, Thierry Monediere and Kevin Elis.

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**Abstract**— This article presents a new concept of compact circularly polarized X-Band [8-8.4GHz] antenna for the 3U-CubeSat platforms. Despite the integration constraints on the top face of the CubeSat, the design aims at an isoflux radiation pattern. This antenna associates a driving patch antenna and twelve parasitic crossed dipoles, both to minimize the axial ratio in the opening angle  $\theta= \pm 65^\circ$  and to shape the radiation pattern. The patch excitation is carried out by a compact sequential-phase feed microstrip circuit. This antenna is manufactured and measured.

**Index Terms**— nano-satellite, circularly polarized X-band antenna, isoflux, parasitic dipoles, sequential-phase feed circuit.

## I. INTRODUCTION

THIS paper aims to propose a solution for a circularly polarized X-band antenna with a radiation pattern as close as possible to an isoflux coverage. The complexity of the challenge is to integrate the antenna on the upper face (10cm\*10cm) of a “3U” CubeSat platform. This study has been carried out within the framework of a CNES (French space agency) Research and Technology (R&T) program. This development led to a prototype, thus achieving a Technology Readiness Level (TRL) of 3.

The particularity of the CubeSats is the shape and size standardization that make them today the most popular of all nanosats [1]. These platforms are composed of a stack of elementary volumes (10cm\*10cm\*10cm), named “1U”. A 3U CubeSat looks like a parallelepiped of 10cm\*10cm\*30cm, which allows it to be launched with a P-Pod deployer [1]. The CubeSats and their payloads benefit of low cost and short development, making them very attractive as technology test and demonstration platforms in order to limit risks in future missions. These platforms mainly interest the academics (training), the industry (Technical demonstration and technology) and the government space agencies. Applications of Nanosats include high data rate telemetry, observation, scientific payloads, high-resolution still imaging, maritime applications such as ship tracking [2]... Most of these missions require an on-board VHF or S-band quasi isotropic

antenna for sat-to-sat or sat-to-earth communications, but VHF links have moderate capability (low data rate) due to the narrow bandwidth. To enhance the data rate communication for future missions, it appears useful to investigate higher frequencies offering larger bandwidths.

Our work focuses on the X band to establish a high data rate downlink with Earth. For this purpose, a high-gain base station beam-steering antenna should be used for the LEO nanosat tracking. Therefore, an antenna payload having an ideally isoflux radiation pattern will offer the longer visibility time from the base station. In the context of the CubeSat, the EIRP is limited by the low available RF power (< 2Watt on 3U CubeSat platforms) and the large radiation coverage required by this future mission (isoflux antenna). For this reason, the antenna must offer both good radiation efficiency and good circular polarization over all directions.

The antenna proposed in this paper is designed to be integrated on the 10cm\*10cm square face of 3U CubeSat. The antenna is compact and thin enough to be compatible with P-Pod launcher. The French space agency (CNES) provided specifications presented in table I.

TABLE I  
ANTENNA SPECIFICATIONS

Parameters	Specifications
Frequency Band (GHz)	X-band [8.0 - 8.4 GHz]
Return Loss (dB)	< -20 dB
Polarization	RHCP
Limit of Coverage	$\theta = 65^\circ$
Minimum Gain	0 dBi
Radiation Pattern	Isoflux (if possible)
Axial Ratio (dB)	< 3dB in the opening angle $\theta = \pm 65^\circ$
Antenna Dimensions	Footprint= 9cm*9cm Thickness < 9mm outside the Satellite
Admissible RF Power	2Watt

A recent paper about the antenna developments for CubeSats [3] shows that most of antennas are designed for VHF or S bands, and only a few antenna concepts are intended for X band. Among these developments, a directive patch antenna array is proposed in [4], a reflectarray in [5] and a large deployable antenna for SAR systems in [6]. All these antennas are high-gain ones and cannot meet our specifications. In the context of our project, two interesting new developments for CubeSat have explored a miniaturized helix [7] and an EBG Matrix antenna [8] to radiate the isoflux diagram, and our work proposes an alternative to these designs.

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The main challenge of this project is to integrate a high quality circularly polarized isoflux X-band antenna on top of the 10cm\*10cm square earth-face of 3U CubeSat. The antenna families known to meet these requirements are the quadrifilar helix antenna [9] and the choke horn antenna [10]. Unfortunately, the former is more than one wavelength high [11], and the aperture diameter of the latter is greater than ten wavelengths, which is incompatible with our integration requirements. Two other original solutions able to produce isoflux radiation patterns have been studied in the literature: one uses metasurfaces [12] and the other is a slots array fed by the radial mode of a planar waveguide [13]. Again, both solutions are too large for CubeSats.

This letter presents the complete design, realization and measurements of a new antenna which uses parasitic elements for complying with the radiation specifications and the integration requirements.

## II. SUMMARY OF THE ANTENNA PREVIOUS DESIGN

This work refers to the previous general design we published in [14]. We give readers a brief reminder of the antenna conception and the simulated electromagnetic properties. The principle of this antenna is to associate a sequential rotation phase shift driving a patch antenna with a set of parasitic crossed dipoles. These dipoles are used for maximizing the gain and the axial ratio in the opening angle  $\pm 65^\circ$  [14]. The antenna architecture shown in Fig. 1 was optimized to comply with an integration footprint on top the 10cm x 10cm upper face of the 3U CubeSats. It is composed of a circularly polarized patch antenna connected to a compact sequential-phase feed microstrip circuit. This circuit is made up of one oversized  $180^\circ$  hybrid ring coupler and two  $90^\circ$  hybrid couplers, which are folded to fit inside the  $180^\circ$  ring coupler. The patch and the circuit are printed on two stacked RO4003c substrates, creating a buried ground plane between the patch and the couplers' board. The patch antenna is connected to the couplers through four via holes crossing the buried ground plane. The resulting microstrip assembly is placed on a metal cylinder (13mm high and 33mm in diameter), which is surrounded by twelve parasitic crossed dipoles placed on a 45mm-diameter circle, printed on both sides of a second RO4003c 1.524mm-thick substrate. Interactions with the set of crossed dipoles were optimized using the method published in [15] and [16] which has been improved to deal with circular polarization. The method solves reactive loads that must be connected to the dipoles in order to meet the radiation objectives (both diagram shaping and polarization). For right-handed circular polarization, the optimization leads to reactive functions different for the dipoles printed on the upper face and the ones on the back face (crossed dipoles). These reactive functions can be emulated by adjusting the lengths and the gaps of the dipoles. Therefore, the six dipoles that are printed on the upper face of the substrate are 0.8mm wide, 11.7mm long, with a 0.5mm gap. The other six dipoles printed on the back are short-circuited ( $\text{gap}=0$ ), they are 12.3mm long and 0.8mm wide. This second substrate (printed parasitic dipoles) lies 1.5mm above 9.5mm-deep concentric corrugations (Fig. 1). These corrugations both weaken the surface currents and forbid the possible cavity

resonances (the ground plane which is set below the upper face of the nanosat forms an open cylindrical cavity). The antenna assembly thickness is 15mm and only 7.5mm exceed the upper face of the platform. The radiation performances simulated in [14] are recalled: for theta varying from  $0^\circ$  to  $+60^\circ$  the radiation pattern is not isoflux but the gain is always greater than 0.4dBi; on the other hand the axial ratio stays below 2.5dB whatever the radiation direction. The return loss simulated with numerical waveguide ports used as matched terminations is lower than -19dB. This antenna preliminary design fulfills the needs over the entire frequency bandwidth.

Fig. 2 illustrates the contribution of the parasitic dipoles to the radiation optimization: the dipoles increase the gain by more than 1dB at theta  $60^\circ$  and the axial ratio is reduced over a wide aperture angle.

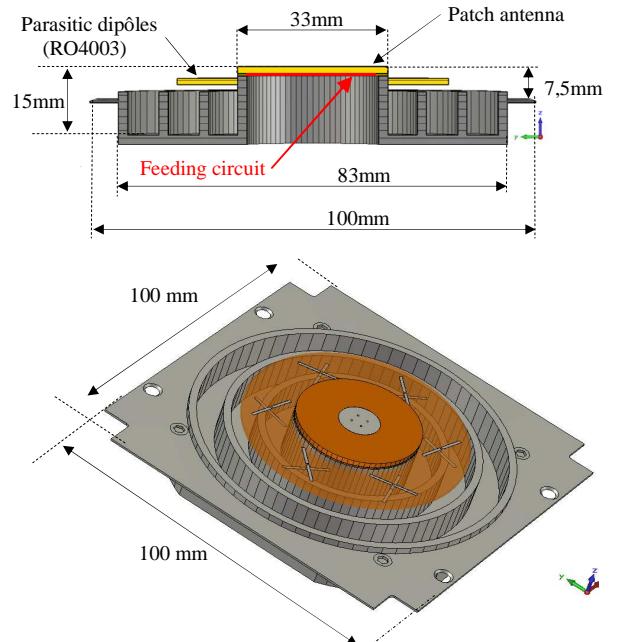


Fig. 1. Antenna design with parasitic crossed dipoles

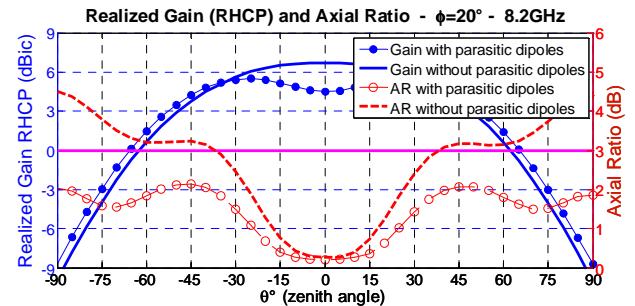


Fig. 2. Demonstration of the parasitic dipoles effects on the radiation pattern and axial ratio (simulations) – plane  $\phi=20^\circ$  - freq=8.2GHz

## III. DESIGN FINALIZATION

The design method used for the antenna conception was presented in [14], and will not be recalled here. However the circuit is now completed by realistic terminations. The numerical waveguide ports used for the simulation are changed into lumped  $50\Omega$  resistors and a coaxial cable is

connected at the input of the circuit. Two different coplanar-to-microstrip transitions are optimized to receive the  $50\Omega$  0402-SMD resistors [17] and the “UT-047” semi-rigid coaxial cable (1.19mm in diameter). The circuit drawing including the coaxial cable and the SMD resistors is presented in Fig. 3. A simulation of this whole circuit confirmed the performances simulated without the physical terminations (in [14]). The magnitude balance is less than 0.2dB between the four patch feeding probes and the maximum phase error between two adjacent feeding probes (via holes 0.5mm in diameter) is about  $2^\circ$  over the frequency bandwidth.

The return loss of the antenna is shown in Fig. 4. We can see that the return loss of circuit alone [14] (pink curve) is slightly degraded by the coaxial transition (black curve). The return loss of the whole circuit with the SMD resistors and the coaxial cable is higher than -13dB over [8 – 8.4GHz] whereas it was -19dB for the circuit terminated by numerical waveguide ports (from [14]). The patch active return loss (sequential-phase feed) reaches -11dB at 8GHz and -9dB at 8.4GHz (Fig. 4), these values will degrade the antenna radiation efficiency.

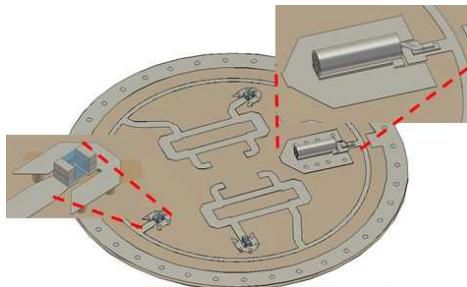


Fig. 3. The sequential-phase feed circuit with the  $50\Omega$  SMD resistors and the coaxial connector.

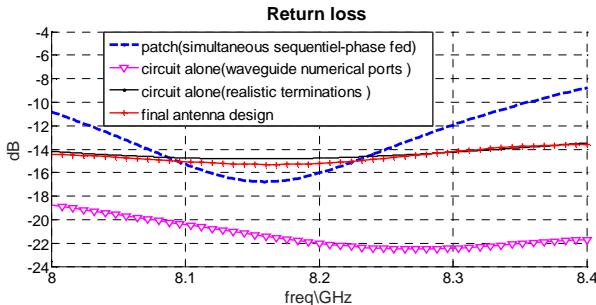


Fig. 4. Return loss at different steps in the antenna design (simulations)

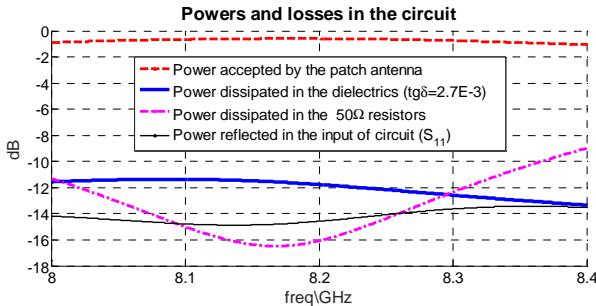


Fig. 5. Power and losses in the circuit (simulation)

The circuit losses are analyzed in Fig. 5. The circuit dielectric losses are around 0.3dB (dissipated power is less than -11dB). By design, powers reflected by the patch are dissipated in the  $50\Omega$  resistors, and the total lost power is lower than -10dB (-9dB at 8.4GHz). Finally, we evaluate the total efficiency of the simulated sequential-phase feed microstrip circuit to be around -1dB.

#### IV. REALIZATION AND MEASUREMENTS

The different parts of the antenna have been manufactured and put together. The patch antenna and the sequential-phase feed circuit are printed on two RO4003c substrates ( $\epsilon_r=3.55$ ,  $\text{tg}\delta=2.7\text{E}-3$ ). The 1.524mm-thick patch substrate is stacked with the 0.406mm-thick feeding circuit. Connections between circuit and patch are four 0.5mm metallized via holes. The upper and lower sides of this circularly polarized driving patch antenna are shown in Fig. 6 (right). Fig. 7 proposes a view of the coaxial cable connecting the circuit and the SMD resistors [17] that are soldered on the optimized terminations. The parasitic crossed dipoles are printed on both sides of a second 1.524mm thick RO4003c substrate (Fig. 6, left). For the measurements, the assembled antenna is finally set up on a 3U-CubeSat platform (Fig. 8).

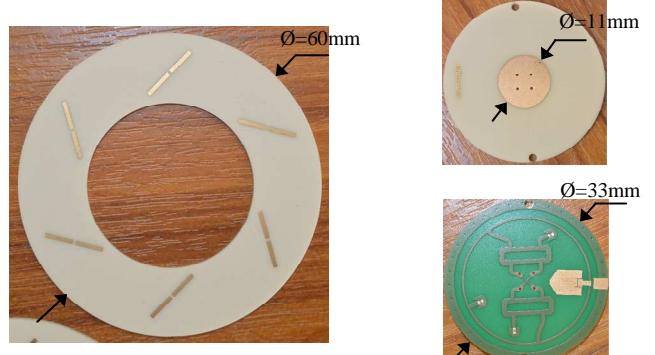


Fig. 6. Parasitic dipoles are printed on both sides of RO4003c substrate (left) - The patch antenna is printed on the 2 layers assembly and the sequential-phase feed circuit is printed backside (right)



Fig. 7. SMD components and coaxial cable are soldered on the circuit.



Fig. 8. The manufactured antenna is integrated on top of a 3U platform

The measured realized gain is plotted at 8.2GHz for  $\phi=0^\circ$  and  $\phi=45^\circ$ , and these plots are compared with simulations (Fig. 9 and Fig. 10). The gain is -0.4dBi at  $\theta = 60^\circ$ , which is 0.8dB lower than the simulation and indicates higher losses in PCBs. The cross-polarization discrimination is greater than 15dB for  $\theta$  varying between  $-60^\circ$  and  $+60^\circ$ . The measured axial ratio (plotted in Fig. 11 and Fig. 12) meets the specifications ( $<3$ dBi) from 8 to 8.4GHz and agrees with the simulation. Due to the symmetries in the antenna design, the radiation patterns are assumed to remain unchanged for any cutting plane over the entire frequency band. This expected omni-directivity is assessed in Fig. 13: the gain and axial ratio plotted in azimuth planes at three elevation angles ( $\theta=30^\circ$ ,  $50^\circ$  and  $60^\circ$ ) confirm the omnidirectionality of the manufactured antenna.

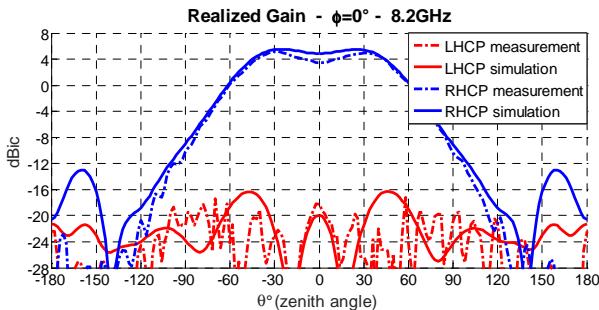


Fig. 9. Measured and simulated realized gain – plane  $\phi=0^\circ$

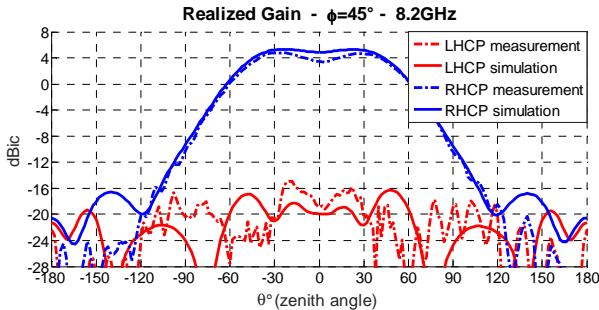


Fig. 10. Measured and simulated realized gain – plane  $\phi=45^\circ$

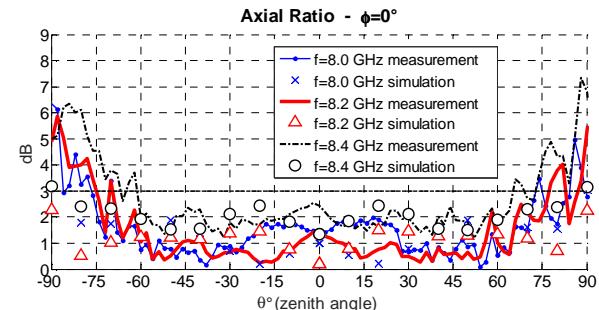


Fig. 11. Measured and simulated axial ratio – plane  $\phi=0^\circ$

The measured antenna return loss (Fig. 14) is lower than -10dB. The antenna is well matched even if there is a small difference between measurement and simulation (Fig. 4) which can be explained by the coaxial cable transition assembly. From Fig. 14, the measured radiation efficiency is about 70% whereas the simulated one is over 80%. This

difference is explained by a possible mismatching between the patch and the circuit. In such a case the reflected power is dissipated in the  $50\Omega$  resistors.

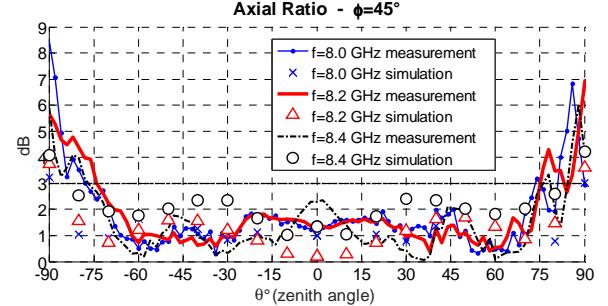


Fig. 12. Measured and simulated axial ratio – plane  $\phi=45^\circ$

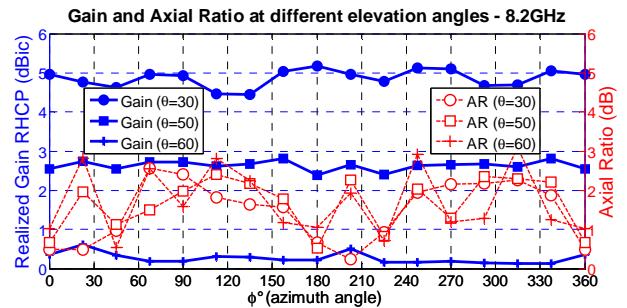


Fig. 13. Measured gain and axial ratio at 3 elevation angles

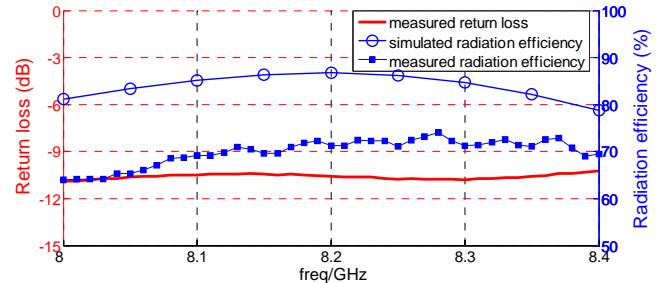


Fig. 14. Measured return loss and radiation efficiency

## V. CONCLUSION

This paper demonstrates the performances of an original X-band circularly polarized antenna developed for integration on the top face of a 3U-CubeSat. The antenna is 15 mm thick with only 7.5mm outside the satellite. The exact isoflux radiation objective could not be achieved due to the platform size yet the maximum gain is at theta 30° and it reaches 0dB at 60°. The minimization of the axial ratio in the opening angle  $\pm 65^\circ$  was carried out by introducing parasitic crossed dipoles. The measured radiation efficiency is about 70%. All the measurements agree with the simulations except for the return loss, which suffer from a defective connector. This new design is a laboratory prototype, and can still be upgraded. A more efficient connector and a wider bandwidth patch would improve the antenna efficiency.

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